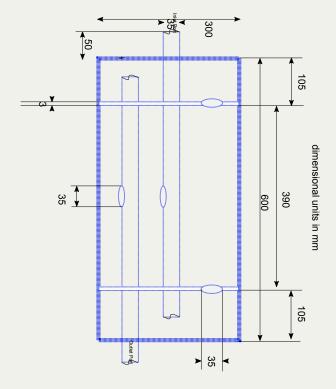
### Multichamber Muffler System

Michael Raba, MSc Candidate at University of Kentucky

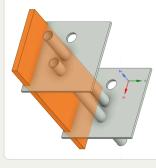
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			Multicomponent Muffler Internal Geometry

#### Dimensions



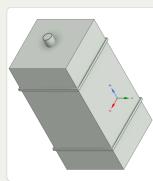
## Schematic Variants for Muffler Subcomponents



Part 1 — Chamber and Baffle

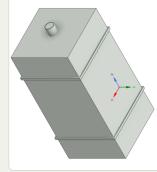
Part 4 — Showing perforates (aimed at fiberglass)

Part 5 — Final Assembly View



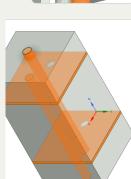
Part 2 — Fluid domain

**-**,



Part 3 — Fiberglass Absorbant (gold)





5

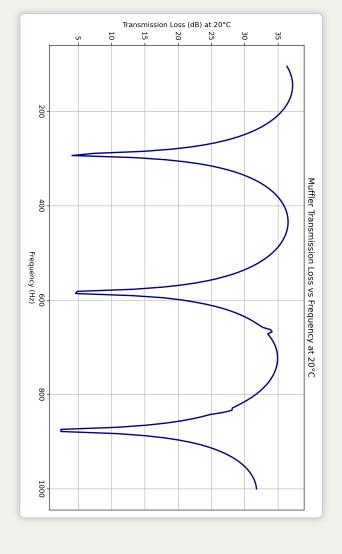


Figure: Transmission Loss curve of the muffler between 5 Hz and 1000 Hz at 20°C.

Simulated Transmission Loss (0–1000 Hz) Simlab model

Transmission Loss vs Frequency

Transmission Loss (dB)

20

200

400

600 Frequency (Hz)

800

1000

40

120 -

140 -

Simlab Simulation

7

### Sidlab and Ansys File Download Center

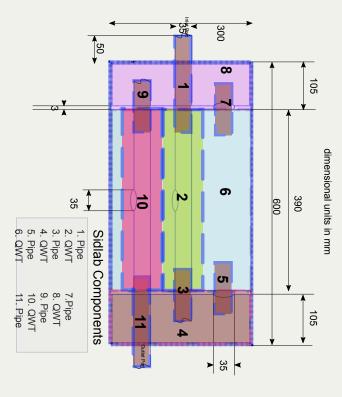
#### SIDLAB Model

- File: Mark3Sid.zip
  Created with: SIDLAB 5.1
  Download SIDLAB File

#### **ANSYS Simulation**

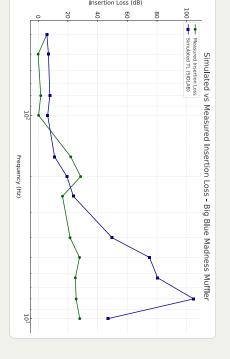
- File: Mark-I-MDF-clearned-data.wbpz
   Greated with: ANSYS 2023 R2
   Download ANSYS File

#### Sidlab Components



### Simulated vs Measured Insertion Loss

### Measured vs Simulated TL



### Insertion Loss Explanation

Insertion Loss (IL) quantifies how much sound is attenuated when a muffler is added to the system.

#### General formula:

$$ext{IL} = 10 \log_{10}\!\left(rac{P_{ ext{baseline}}}{P_{ ext{muffler}}}
ight)$$

Because our data is already in decibels (dB), this simplifies to:

 $IL = Power_{baseline \; (dB)} - Power_{muffler \; (dB)}$ 

#### References

#### Cited Works

1. Munjal ML. Acoustics of Ducts and Mufflers. 2nd ed. Wiley; 2014. ISBN: 9781118443125. https://doi.org/10.1002/9781118443125

2. Dokumacı E. Duct Acoustics: Fundamentals and Applications to Mufflers and Silencers. Cambridge University Press; 2021. ISBN: 9781108840750. https://doi.org/10.1017/9781108840750

Note: These references are foundational texts in muffler and duct acoustics and were consulted for system modeling, schematic development, and transmission loss analysis.

# Anand Model: Viscoelastoplasticity and its Application to Solder Joints

Michael Raba, MSc Candidate at University of Kentucky

Created: 2025-09-15 Mon 12:39

## Constitutive Equations for Hot-Working of Metals

Author: Lallit Anand (1985)

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**DOI:** 10.1016/0749-6419(85)90004-X

One of the foundational papers in thermodynamically consistent viscoplasticity modeling—especially significant in the context of metals subjected to large strains and high temperatures.

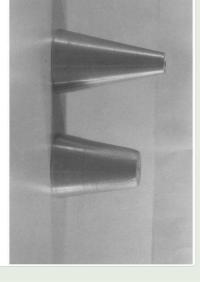


Fig. 25. 1100 aluminum state gradient specimens before and after testing

Animational Assemble of Passicity, Vol. 1, pp. 213-231, 19 Printed in the U.S.A.

0789-6419415 53.00 + .00 © 1985 Pergamon Press Ltd.

#### CONSTITUTIVE EQUATIONS FOR HOT-WORKING OF METALS

LALLIT ANAND

ommunicated by Theoder Lehmann, Ruhr Universität Bochum)

offering the manufacturing of most small groduats. Causal is not assessful ababits of a ton-severalized grocest in the use of superpostes as and impression-specialized contacting expension for the content of the cont

#### I. INTRODUCTION

Howevering is an important processing step during the manufacture of approximately more than eighty-live percent of all metal products. The main features of hot-averting are than metals are deformed into the desired shapes at temperatures in the range of —0.3 through ~0.9 %, where % is the melting temperature in degree televity, and at strain rates in the range of ~10° through ~0.9 %ec. It is to be noted that most hot-working processes are more than mere shape-making operations; an important goal of hot-working is to subject the workpiece to appropriate thermo-mechanical processing shorters which will produce microstructures that optimize the mechanical properties of the product.

The major quantities of metals and alloys are hot-worked under interrupted nonitothermal conditions. The principies of the physical metallurgy of such deformation processing are now well recognized, e.g., Jones et al. [1996], Statasta self-Ol Texastr [1971], McQueras a Joses [1975], and Stataste [1978]. During a deformation pass, the stress is found to be a strong function of the strain art, temperature, and the deformation tends to be a strong function of the strain art, temperature, and the deformation tends to be counteracted by dynamic recovery processes. These recovery processes result in a tearrangement and annihilation of dislocations in such a manner that as the strain in a pass increases, the dislocations tend to arrange themselves into subgain walls. In some metals and alloys (especially those with a high stacking datal energy, e.g., Al., o-Fe and other ferritia alloys [40] of the self-level and maintained to large strains before feature occurs. In other metals and alloys in which recovery it is study (depecially those metals with low stacking fault energies, e.g., Ni, y-Fe and other austernitic

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Case Study: Wang (2001) Apply to Solder

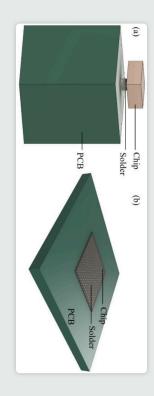
G. Z. Wang pulsary supplications of the Control of	Applying Anand Model to Represent the Viscoplast Deformation Behavior of Alloys	Anand N it the Vis ion Beha	Applying Anand Model to Represent the Viscoplastic Deformation Behavior of Solder Alloys	
K. Becker University of Applied Science Bellyon, Bellyon SECEL (Secesion Provided Control Security Control Provided Control ACRIS, University of Perforance Science of Security Control Security of Security Control Security Contr	A suited stouplestic trocket diplomation in parameter of the coast of Table 2018. The area of the coast of th	constitutive less, the debestice for reddern a finalise relations for fi- tractic relations for fi- rer decreasined from a sirred analised Assaul- fy-come plants; flow and the Assaul model or of reddern at high ha- cer atmentation of the	I will tricipalite committee lite, the Least mode, the applied to approve the freedom the America of America All Committee and C	
soni, suspensuse the transfore experienced by IC pack- lassembles in service cases progressive damage in value restaudy, this damage accumulation beyond certain limits		which accounts for the case energy and for the Ques et al. [16] emple	which accounts for the measured stress-dependence of the activa- tion energy and for the Bassalinger effect exhibited by the solder. Quin et al. [16] employed the back area to describe the transferr states of a stress/stress come in a suited constitution model for	
the electrical failures. One of the major gratin of therms cal analysis in the electronics industry is to be able as the attenuatata responses of the solder joint and that is utiliability in service. In order to gain accurate simula-	njer puth of themso- nery is to be able to solder joint and then pain accurate simula-	in had stake Some raphyol for Badeo rate variable to spen other variable to	tin-had solder. Some rootts (Skipe et al. [11]) Ma et al. [12]) employed the Bother-Prison constitutor relations which are a state variable to appreciat the internal includes attention for the solder with control control-tion. Herwork, some parameter for for	
mand principal, reason containers reasons or on- reason was about the principal and the first for the least Softh at more conception or another the the		the saided models and examine the common and some to	flower unafied models are empirical and dependent on temperature and carele case, remaining to complete calculations of the stones strain empowers and some sentering predictions from the experiments.	
int and the thermally activated strains imposed on it due sexual expansion mountait between the materials gives complex deformation behavior. This deformation behav-		fined as a user-define sale-dependent stress- grams (Busses et al. )	fixed as a vary-defined advocation code to represent the nonlinear rate dependent stero-crasin relations in some finite element per- pares (blacos et al. 15). Quan et al. [10]. Such a work is often	
therefore inclusive characteristics, producing strain the dynamic recovery, and in many instances dynamic to		mer marcial submode mer marcial submode mady available in our	ready available in current communicial finite element codes, e.g.,	
with, presently, reventions such as comp and save in phenomena of mancials working in high homologue to previous.		the model, which is a the ANSYS code. In a	the cross's in Analysis (receive in a [13]); A making consists to the condition of the cond	
has already been a great deal of effort applied to reason- evimental data and constitutive models for this material vious researchers (Darwaux and Banerji [1]; Weinhol	fert applied to reason- deb for this material. Basorji [1]: Weinbel	packaging, the manerial par must be determined in prior. The objective of this paper	pakhajásy, the manoial parameters of the constitutive aductions much be determined in press.  The objective of this paper is to obtain the manoial parameters.	
j. Kaidopp and Murzy [3] presented cotonive coperimen as include based solders. There are also some constitution for solder alloys, ranging from classic plants: model (e.g.		of the Assaul model for separated elisate plants parameters with visco	of the Assalt model for solders from opportunistal results and the reputated bishelphane comp constitution in thirtiests. The unserial parameters with viscoplantic constitution printings for solders are	
Object relation) using empirical streaments corre- fiber (4) to a purely phenomenological model where the nodest and stan-andependent delicents are artificially a Sanhara [5]. Pao et al. [4]. Knocht and Fay [7]). Some		not to simulate the nate behavior, and at making for companion	used to simulate the sensity-state energy behavior, constant strain can behavior, and strawa varian toposous under thermal cyclic bunding for comparison and verification. Some discussions on rev- mementation of using unified Auntal model in the faithr between internation of straing unified Auntal model in the faithr between the contract of decreased production of solidity wares done presented.	
sperbolic size errop, etc.) have been applied to the size of energ data. However, from the viewpoint of continuum cs, the time-dependent and sime-independent inelastic		Model Formulation		
et prounted to arise from similar mechanisms due to dis marion. So, a unified framework for stooplantic behavior canaritals in highly desired. However, those cold land ords smill some Seman sensember Disease et al. D.H. however,		<ol> <li>The Anamal Model: A for large, isotropic, viscoplastic formations is the single-scalar by Anamal and Brown (Anamal</li> </ol>	The Annual Model. A simple set of constitutive equations for large, isotropic, viscoplarie deformations but small elastic de- formations is the single scalar insumal variable annual proposed by Annual and Brown (Annual [14], Brown et al. [15]). There are by Annual and Brown (Annual [14]).	
and by the Thermote and Phasanic Packaging Stokins for publication in As. of ELECTRONIC PRESCRIPTION Measuring montred by the EPPS On the Assentian Edwar Vacidate Pres.		we basic features in 6 capticit yield condition therefore strain is assume although at low stress	two basic features in this Annual model. First, this model needs no explicit yield condition and no loading/ministing citization. The explicit point counts to take place and all measure strates values, admosph at low streams the sate of placin: then may be immen-	
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urce: Wang, C. H. (2001). "A Unified pp-Plasticity Model for Solder Alloys." Doi: 10.1115/1.1371781

Applies Anand's unified viscoplastic framework to model solder behavior.

Why Wang's Paper Matters

- Anand's model can be reduced and fitted from experiments.
- transition the theory into engineering-scale implementation.
- Targets solder joints in microelectronic packages (chip on PCB, soldered connections).



## Comparing Anand Model Predictions at Two Strain Rates

#### **Observed Behavior**

Recovery negligible → pronounced hardening

High strain rate → higher stress Top Graph (a):  $\dot{\varepsilon} = 10^{-2} \, {\rm s}^{-1}$ 

- Recovery and creep effects more significant

Model Accuracy: Lines = model prediction, X =

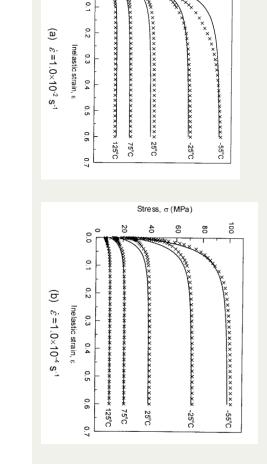
experimental data

- Lower strain rate → lower stress at same strain Bottom Graph (b):  $\dot{\varepsilon}=10^{-4}\,\mathrm{s}^{-1}$
- "At lower strain rates, recovery dominates... the Key Insights from Wang (2001)

stress levels off early."

Anand's model smoothly captures strain-rate and "At high strain rates, hardening dominates, and the stress grows continuously."

temperature dependence of solder materials.



Stress, σ (MPa)

80 60

20 40

0.2

0.4

Inelastic strain, ε

100 120

15

## Main Equations of Wang's Anand-Type Viscoplastic Model

### Flow Rule (Plastic Strain Rate)

 $\dot{\varepsilon}^p = A \exp \biggl( - \frac{Q}{RT} \biggr) \biggl[ \sinh \biggl( \frac{j\sigma}{s} \biggr) \biggr]^{1/m}$ 

$$\dot{arepsilon}^p = A \exp \left( -rac{\epsilon_d}{RT} 
ight) \left[ \sinh \left( rac{j\sigma}{s} 
ight) 
ight]$$

• Plastic strain rate increases with stress and

No explicit yield surface; flow occurs at all nonzero stresses. temperature.

### Deformation Resistance Saturation $s^*$

$$s^* = \hat{s} \left(rac{\dot{arepsilon}^p}{A} ext{exp} \left(rac{Q}{RT}
ight)
ight)^n$$

- ullet Defines the steady-state value that s evolves
- Depends on strain rate and temperature.

### Evolution of Deformation Resistance $\boldsymbol{s}$

$$\dot{s} = h_0 \left| 1 - rac{s}{s^*} 
ight|^a ext{sign} \left( 1 - rac{s}{s^*} 
ight) \dot{arepsilon}^p$$

- Describes dynamic hardening and softening of the material.
- s evolves depending on proximity to  $s^{\ast}$  and flow activity.
- Note: Constants  $A,Q,m,j,h_0,\hat{s},n,a$  are material-specific and fitted to experimental creep/strain rate data.

#### **Image Reference**

## Values are from correspond to 60Sn40Pb solder

parameters used in Anand's model:

- $S_0$ : Initial deformation resistance
- Q/R: Activation energy over gas constant A: Pre-exponential factor for flow rate

m: Strain rate sensitivity of stress

 $\xi$ : Multiplier of stress inside sinh

h<sub>0</sub>: Hardening/softening constant

- $Q/R=10830~\mathrm{K}$
- $\xi = 11$  $A = 1.49 \times 10^7 \text{ s}^{-1}$

**Numerical Values** 

 $S_0=5.633 imes 10^7$  Pa

- m = 0.303
- $h_0=2.6408 imes10^9$  Pa
- $\hat{s}=8.042 imes 10^7\,\mathsf{Pa}$ n=0.0231
- a = 1.34
- These constants match Wang's paper for modeling 60Sn40Pb viscoplasticity.

a: Strain rate sensitivity of hardening or ŝ: Coefficient for saturation stressn: Strain rate sensitivity of saturation

softening

17

## Forward Euler Explicit time integration scheme Pseudocode

#### Initialization

- Material constants:  $A,Q/R,j,m,h_0,\hat{s},n,a,E$  Strain rate:  $\dot{\varepsilon}$
- Temperature set:  $\{T_i\}$
- Set:  $\varepsilon^p(0) = 0$ ,  $s(0) = \hat{s}$

Time Evolution Loop

1. 
$$arepsilon_{ ext{total}}(t) = arepsilon t$$
2.  $\sigma_{ ext{total}} = E(arepsilon_{t})$ 

1. 
$$arepsilon_{ ext{total}}(t) = \dot{arepsilon} t$$
  
2.  $\sigma_{ ext{trial}} = E(arepsilon_{ ext{total}} - arepsilon^p)$ 

3. Compute 
$$x=rac{j\sigma}{s}$$

4. Approximate  $\sinh(x)$  (linearize if  $|x|\ll 1$ ) 5.  $\dot{\varepsilon}^p=Ae^{-Q/RT}(\sinh(x))^{1/m}$ 

Plastic Flow & Resistance Evolution

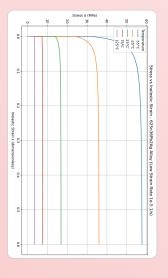
6. 
$$s^* = \hat{s} \left( \frac{arepsilon^p}{A} e^{Q/RT} \right)^n$$

7. 
$$\dot{s}=h_0|1-\frac{s}{s^*}|^a \mathrm{sign}\left(1-\frac{s}{s^*}\right)\dot{\varepsilon}^p$$
8. Update:  $\varepsilon^p(t+\Delta t)=\varepsilon^p(t)+\dot{\varepsilon}^p\Delta t$ 
9. Update:  $s(t+\Delta t)=s(t)+\dot{s}\Delta t$ 

9. Update: 
$$s(t+\Delta t)=s(t)+\dot{s}\Delta$$
10. Record  $(arepsilon_{
m total},\sigma_{
m trial})$ 

#### Termination

- Stop when  $\varepsilon_{\mathrm{total}} \geq \varepsilon_{\mathrm{max}}$
- Plot  $\sigma$  vs  $\varepsilon$  for all  $T_i$



### Forward Euler Scheme for Anand Model

```
# Mintrial constants for 625n36Pb2dg solder alloy A = 2.0486 of K

J = 13

B = 11200 d K

S = 1230 d K

S = 1230 d Gamentionless

S = 1230 d Gamentionless

S = 1230 d F

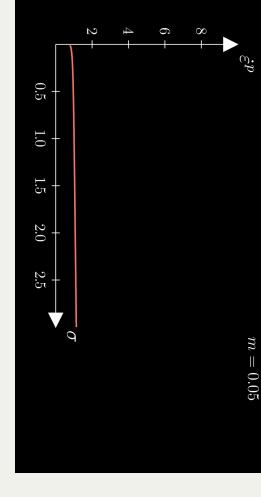
S = 12470 d F

S
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             # Temperatures in Kelvin
T_C = [-55, -25 25 75 125]
T_list = [T + 273.15 for T in T_C]
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                import numpy as np
import matplotlib.pyplot as plt
from scipy.integrate import solve_ivp
| dep_r = np maximum(sinb_x, to_1)|
| dep_r = n = np maximum(sinb_x, to_1)|
| s_star = s_bat * (do_r / 1) * sinb_**(t/n)|
| di = hd * np.max(t = s/s_star)**a * np.stgn(t - s/s_star) * dep_p
| return (dep_p, ds)|
```

### Strain rate sensitivity of stress m

- As  $m \to 0$ , rate insensitive (yield) As  $m \to 1$ , small stress change causes big change in strain rate

## Anand Flow Law: Varying \$m\$



Tensorial Flow Rule (directional form)

$$\mathbf{D}^p = \dot{\epsilon}^p \left(rac{3}{2} rac{\mathbf{T}'}{ar{\sigma}}
ight)$$

Equivalent Stress Definition

$$ar{\sigma}=\sqrt{rac{3}{2}\mathbf{T}':\mathbf{T}'}$$

Plastic Strain Rate (magnitude form)

$$\dot{\epsilon}^p = A \exp \left( -rac{Q}{R heta} 
ight) \left[ \sinh \left( \xi rac{ar{\sigma}}{s} 
ight) 
ight]^{1/m}$$

Full Flow Rule with Hyperbolic Sine

$$egin{aligned} \mathbf{D}^p &= A \exp\left(-rac{Q}{R heta}
ight) \left[\sinh\left(\xirac{ar{\sigma}}{s}
ight)
ight]^{1/m} \left(rac{3}{2}rac{\mathbf{T}'}{ar{\sigma}}
ight), \ &= \dot{\gamma}^p \left(rac{\mathbf{ ilde{T}}'}{2ar{ au}}
ight), \quad ar{ au} = \left\{rac{1}{2} ext{tr}(\mathbf{ ilde{T}}'^2)
ight\}^{1/2} \end{aligned}$$

- Direction given by T'.
- Magnitude determined by hyperbolic sine based on  $\bar{\sigma}/s$ .
- $ar{ au}$  represents the effective shear stress computed from deviatoric stress
- $ar{\sigma}=\sqrt{rac{3}{2}}{f T}':{f T}'$  is the von Mises Equivalent stress, but is formally defined without yield point
- Summary: Full flow = direction × magnitude.

ss Evolution Equation (Rate form of Hooke's Law)

$$\mathbf{T} = \mathbb{L}\left[\mathbf{D} - \mathbf{D}^p
ight] - \mathbf{\Pi}\dot{ heta}$$

rate-form Hooke's law for finite deformation ticity, with frame-indifference enforced through the Jaumann rate.)

Jaumann Rate Definition

$$\mathbf{T} = \dot{\mathbf{T}} - \mathbf{WT} + \mathbf{TW}$$

Summary:

Stress rate follows Jaumann derivative to ensure frame indifference. Elastic response governed by isotropic fourth-order tensor L.

Thermal expansion introduces additional stress through  $\mathbf{\Pi} heta$ 

### Material Tensors and Operators

- $\mathbb{L} = 2\mu \mathbf{I} + \left(\kappa \frac{2}{3}\mu\right)\mathbf{1}\otimes\mathbf{1}$  isotropic elasticity tensor
- LD represents how instantaneous strain rates generate stresses according to the elastic material's stiffness properties
- stiffness properties.  $\mu = \mu(\theta)$ ,  $\kappa = \kappa(\theta)$  temperature-dependent moduli
- $\Pi = (3\alpha\kappa)\mathbf{1}$  stress-temperature coupling
- $\alpha = \alpha(\theta)$  thermal expansion coefficient
- $\alpha = \alpha(\sigma)$  the final expansion coefficients  $\mathbf{D} = \operatorname{sym}(\nabla \mathbf{v})$  stretching tensor
- $\mathbf{W} = \operatorname{skew}(\nabla \mathbf{v})$  spin tensor
- I = fourth-order identity tensor
- 1 = second-order identity tensor

### Stress Evolution and Thermal Effects

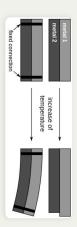
In the stress evolution equation,

$$\mathbf{T} = \mathbb{L}\left[\mathbf{D} - \mathbf{D}^p
ight] - \mathbf{\Pi}\dot{ heta},$$

erm  $\mathbf{\Pi} \hat{\theta}$  represents the stress change that would occur pure thermal expansion alone, without any mechanical loading.

### Why Subtract the Thermal Term?

- Thermal expansion creates strain even without external forces.
- Without subtracting  $\Pi\theta$ , the model would falsely attribute thermal strain as mechanical stress.
- Subtracting isolates the true mechanical response from thermal effects.



- Thermal expansion induces strain without force.
- Subtracting  $\Pi\dot{ heta}$  ensures only mechanical strains generate stresses.
- Summary: This keeps the constitutive model physically accurate during heating and cooling

### Relaxed (Intermediate) Configuration

### Context for the Relaxed Configuration

All thermodynamic potentials, internal variables, and elastic deformations. removing plastic deformations but before applying new from recoverable elastic effects. It is introduced to separate permanent plastic effects The relaxed configuration represents the material after

evolution laws are defined relative to this frame.

for measuring elastic strain  $E^{e}$  and computing The relaxed state provides a clean, natural reference

What Happens in the Relaxed Configuration?

- The elastic deformation gradient  $F^{e}$  is measured from Elastic strain measures like  $C^e$  and  $E^e$  are defined in the relaxed state to the current deformed state.
- this configuration.
- The Kirchhoff stress  $\widetilde{\mathbf{T}}$  is naturally associated with the relaxed volume.
- Plastic flow is accounted for separately through the plastic velocity gradient  $\mathbf{L}^p$ .

#### Summary:

The relaxed configuration isolates elastic responses cleanly, enabling proper definition of thermodynamics and plastic evolution laws.

### Kinematics in the Relaxed Configuration

Elastic deformation gradient:

$$F = F^e F^p \quad \Rightarrow \quad F^e = F F^{p-1}$$

Elastic right Cauchy-Green tensor:

$$C^e = F^{eT} F^e$$

Elastic Green–Lagrange strain tensor:

$$E^e=rac{1}{2}(C^e-I)$$

### Stress and Power Quantities

Kirchhoff stress (weighted Cauchy stress):

$$\widetilde{\mathbf{T}} = (\det F)\mathbf{T}$$

Stress power split:

$$\dot{\omega}=\dot{\omega}^e+\dot{\omega}^p$$

 $\dot{\omega}^e = \widetilde{\mathbf{T}} : \dot{E}^e$ 

$$ec{\mathbf{r}}:\dot{E}^e$$
 ,  $\dot{\omega}^p=(C^e\widetilde{\mathbf{T}}):\mathbf{L}^p$ 

#### Summary:

- Elastic kinematics and stress measures are formulated relative to the relaxed configuration, cleanly separating plastic and elastic contributions.
- Stress Power Split allows Anand to cleanly isolate plastic dissipation from elastic storage
- Green-Lagrange strain tensor  $E^e$  is used because it symmetrically captures nonlinear elastic strain relative to the relaxed configuration
- The right Cauchy-Green tensor  $C^e = F^{e^T}F^e$  is required as an intermediate to compute  $E^e$  from the elastic deformation gradient  $F^{e}$  without referencing spatial coordinates

Thermodynamic Separation

### 1. Start with Total Dissipation:

$$\mathcal{D} = \dot{\omega} - \dot{\psi} \geq 0$$

where 
$$\dot{\omega}=\widehat{\mathbf{T}}:\dot{\mathbf{E}}^e+(\mathbf{C}^e\widehat{\mathbf{T}}):\mathbf{L}^p$$
 2. Split Stress Power:

 $\dot{\omega}=\dot{\omega}^e+\dot{\omega}^p$ 

• 
$$\dot{\omega}^p = (\mathbf{C}^e \widehat{\mathbf{T}}) : \mathbf{L}$$

•  $\dot{\omega}^e = \widehat{\mathbf{T}} : \dot{\mathbf{E}}^e$ •  $\dot{\omega}^p = (\mathbf{C}^e \widehat{\mathbf{T}}) : \mathbf{L}^p$ 3. Group Terms with  $\dot{\psi}$ :

$$(\dot{\omega}^e - \dot{\psi}) + \dot{\omega}^p \geq 0$$

4. Apply Elastic Energy Consistency:

$$\dot{\omega}^e - \dot{\psi} = 0 \quad \Rightarrow \quad \dot{\omega}^p \ge 0$$

#### Key Physical Insights

- Elastic deformations are recoverable and do not cause entropy production.
- All dissipation stems from the plastic flow:  $\dot{\omega}^p$
- Plastic work increases entropy and governs viscoplastic evolution.

#### Summary:

The stress power split ensures that the second law is satisfied by assigning dissipation solely to irreversible processes.

#### Reference Configuration

The free energy  $\psi$  is defined relative to the reference

Framework in the Reference Configuration

configuration. State variables like  $E^e, \theta, \bar{g}, \bar{\mathbf{B}}, s$  are used as arguments of  $\psi$ .

arguments of  $\psi$ . Stress is expressed using the second Piola–Kirchhoff

tensor S. Dissipation inequality, stress–strain relations, and evolution laws are all written in reference variables. Mass density  $\rho_0$  from the reference configuration normalizes all terms.

Key Equations in the Reference Frame

Free energy:

$$\psi=\psi(E^e, heta,ar{g},ar{{f B}},s)$$

Dissipation inequality:

$$\dot{\psi} + \eta \dot{ heta} - 
ho_0^{-1} \mathbf{S} : \dot{E} + (
ho_0 heta)^{-1} \mathbf{q}_0 \cdot \mathbf{g}_0 \leq 0$$

Constitutive relation:

$$\mathbf{S} = 
ho_0 rac{\partial \psi}{\partial E^e}$$

#### Summary:

 In the reference configuration, all energy storage, stress updates, and internal variable evolution are formulated with reference-frame quantities for consistency and objectivity.

### Thermodynamic Quantities

$$\psi = \epsilon - \theta \eta$$

Reduced dissipation inequality:

$$[\dot{\psi} + \eta \dot{ heta} - 
ho^{-1} \mathbf{T} : \mathbf{L} + (
ho heta)^{-1} \mathbf{q} \cdot \mathbf{g} \leq 0$$

State variables:

$$\{E^e,\theta,\bar{g},\bar{\mathbf{B}},s\}$$

with  $E^e$  as elastic strain and s as internal resistance.

### Stress Power and Kirchhoff Stress

Stress power per relaxed volume:

$$\dot{\omega} = \left(rac{
ho_0}{
ho}
ight) \mathbf{T}: \mathbf{L}$$

Weighted Cauchy (Kirchhoff) stress:

$$\widetilde{\mathbf{T}} = (\det F)\mathbf{T}$$
 or  $\widetilde{\mathbf{T}} = \left(\frac{\rho_0}{\rho}\right)\mathbf{T}$ 

Decomposition of stress power:

 $\dot{\omega} = \dot{\omega}^e + \dot{\omega}^p$ 

$$\mathbf{F}[\mathbf{T}]$$
 or  $\mathbf{T} = \left(\frac{\mathbf{T}}{\rho}\right)$ 

$$\dot{\omega}^e = \widetilde{\mathbf{T}} : \dot{E}^e, \quad \dot{\omega}^p = (C^e \widetilde{\mathbf{T}}) : \mathbf{L}^p$$

$$\omega^{-}\equiv \mathbf{T}:E\;,\quad \omega^{c}\equiv (C^{-}\mathbf{T})$$

$$\omega^c = \mathbf{T} : E \;, \quad \omega^{\nu} = (C^c \mathbf{T}) :$$

Summary:

## POD Analysis of Turbulent Pipe Flow

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Created: 2025-09-15 Mon 12:39

1. Code Execution and Layout

#### 1.1. Layout

- b7.m
   initSpectral.m
   reads in binary files, takes eg m-fft
   → initEigs.m

forms corrMat, finds eigenvalues

#### 1.2. Layout 2

- $L \hookrightarrow initPod.m$
- carries out POD calculations (quadrature, multiplication ggf betwen αΦ) according to Papers (Citriniti George 2000 for Classic POD, Hellstrom Smits 2017 for Snapshot POD)
   ∴ timeReconstructFlow.m
- performs 2d reconstruction + plotSkmr (generates 1d radial graph)

#### 1.3. Important Switches

pipe = Pipe(); creates a Pipe Class. As the functions (above) are called, data is stored in sub-structs:

. obj.Caseld - stores properties like Re, rotation number S, experimental flags such as quadrature (simpson/trapezoidal), number of gridpoints, frequently called vectors (rMat  $r=1,\ldots,0.5$ ) 2. obj.pod - eigen data, used for calculating POD 3. obj.solution - computed POD modes 4. obj.plt - plot configuration

2. Equations Used in Code Procedure

#### 2.1. Classic POD Equations

The following equations are used in the above code.

$$\begin{split} &\int_{r'} \mathbf{S}\left(k;m;r,r'\right) \boldsymbol{\Phi}^{(n)}\left(k;m;r'\right) r' \mathrm{d}r' = \lambda^{(n)}(k;m) \boldsymbol{\Phi}^{(n)}\left(k;m;r\right) \\ &\mathbf{S}\left(k;m;r,r'\right) = \lim_{r \to \infty} \frac{1}{r} \int_{0}^{r} \mathbf{u}(k;m;r,t) \mathbf{u}^{*}\left(k;m;r',t\right) \mathrm{d}t \\ &\alpha^{(n)}(k;m;t) = \int_{r} \mathbf{u}(k;m;r,t) \boldsymbol{\Phi}^{(n)^{*}}(k;m;r) r \, \mathrm{d}r \end{split}$$

### 2.2. Classic POD Equations (Fixed)

$$\int_{r'} \frac{1^{1/2} S_{i,j}(r,r';m;f) r^{1/2} \phi_j^{*(n)}(r';m;f) r^{1/2} dr'}{W_{i,j}(r,r';m;f)}$$

$$= \lambda^{(n)}(m,f) r^{1/2} \phi_i^{(n)}(r;m;f)$$

$$\sum_{\lambda^{(n)}(m;f)}^{\lambda^{(n)}(m;f)} \phi_i^{(n)}(r;m;f)$$

$$\alpha_n(m;t) = \int_{r} \mathbf{u}(m;r,t) r^{1/2} \Phi_n^*(m;r) dr$$

## 2.3. Snapshot POD Equations

$$\begin{split} & \lim_{r\to\infty}\frac{1}{\tau}\int_0^{\tau}\mathbf{u}_{\mathrm{T}}(k;m;r,t)\alpha^{(n)^*}(k;m;t)\mathrm{d}t \\ & =\Phi_{\mathrm{T}}^{(n)}(k;m;r)\lambda^{(n)}(k;m) \\ & \mathbf{R}\left(k;m;t,t'\right)=\int_{r}\mathbf{u}(k;m;r,t)\mathbf{u}^*\left(k;m;r,t'\right)r\,\mathrm{d}r \\ & \lim_{r\to\infty}\frac{1}{\tau}\int_0^{\tau}\mathbf{u}_{\mathrm{T}}(k;m;r,t)\alpha^{(n)*}(k;m;t)\,\mathrm{d}t \\ & =\Phi_{\mathrm{T}}^{(n)}(k;m;r)\lambda^{(n)}(k;m). \end{split}$$

## 2.4. Reconstruction

The reconstruction is given by

$$\begin{split} q(\xi,t) - \bar{q}(\xi) &\approx \sum_{j=1}^{r} a_{j}(t)\varphi_{j}(\xi) \Rightarrow \\ q(r,\theta,t;x) &= \bar{q}(r,\theta,t;x) + \sum_{n=1}^{r} \sum_{m=0}^{r} \alpha^{(n)}(m;t)\Phi^{(n)}(r;m;x) \end{split}$$

Since the snapshot pod implementation is not error-free, the reconstruction can only be recovered by writing for factor  $\gg 0$ .

$$q(r,\theta,t;x) = \bar{q}(r,\theta,t;x) + (\mathrm{factor}\;\gamma) \sum_{n=1}^{\infty} \sum_{m=0}^{\alpha^{(n)}} (m;t) \Phi^{(n)}(r;m;x)$$

## 2.5. Reconstruction

In order to reconstruct in code, caseld.fluctuation = 'off'. This is incorrect. The necessary use of (factor  $\gamma$ ) is incorrect

#### 3. Derivation

To derive the questioned equation, consider the integral: 
$$\frac{1}{\tau}\int_0^\tau \mathbf{u}_{\rm T}(k;m;r,t)\alpha^{(n)^*}(k;m;t)dt.$$

Substitute 
$$\mathbf{u}_{\mathrm{T}}$$
 with its expansion: 
$$\frac{1}{\tau} \int_0^{\tau} \left( \sum_t \Phi_{\mathrm{T}}^{(l)}(k;m;r) \alpha^{(l)}(k;m;t) \right) \alpha^{(n)^*}(k;m;t) dt.$$

### 3.1. 4 Derivation

Exchange the order of summation and integration, and apply orthogonality,

$$\sum_{l} \Phi_{\mathrm{T}}^{(l)}(k;m;r) \left(\frac{1}{\tau} \int_{0}^{\tau} \alpha^{(l)}(k;m;t) \alpha^{(n)^{\star}}(k;m;t) dt \right).$$

Due to the orthogonality, namely that  $\alpha^{(n)}$  and  $\alpha^{(p)}$  are uncorrelated

$$\langle a^{(n)} lpha^{(p)} 
angle = \lambda^{(n)} \delta_{np}$$

all terms where  $l \neq n$  will vanish, and there remains only the l=n term,

$$\Phi_{\mathrm{T}}^{(n)}(k;m;r)\left(\frac{1}{\tau}\int_{0}^{\tau}\alpha^{(n)}(k;m;t)\alpha^{(n)^{*}}(k;m;t)dt\right).$$

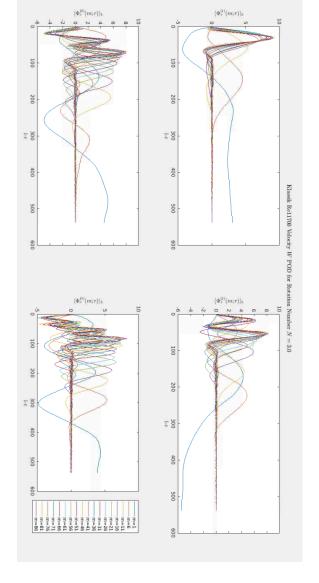
derivation assumes the normalization of modes and their orthogonality, along with the eigenvalue relationship to simplify the original integral into a form that reveals the spatial structure ( $\Phi_{\Gamma}^{(n)}$ ) of each mode scaled by its significance ( $\lambda^{(n)}$ ).

#### 3.2. 6 Derivation

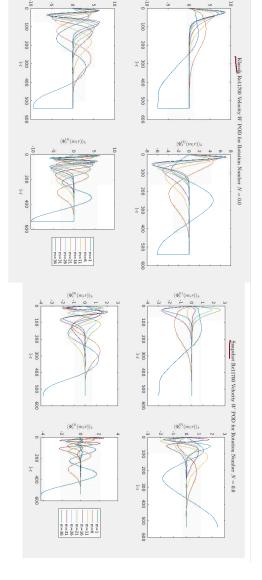
ross-correlation tensor  $\mathbf R$  is defined as  $\mathbf R$   $(k;m;t,t')=\int_r \mathbf u(k;m;r,t)\mathbf u^*\left(k;m;r,t'\right)r\ dr$ . This tensor is now transformed from  $[3r\times 3r']$  to a  $[t\times t']$  tensor. The n POD modes are then constructed as,

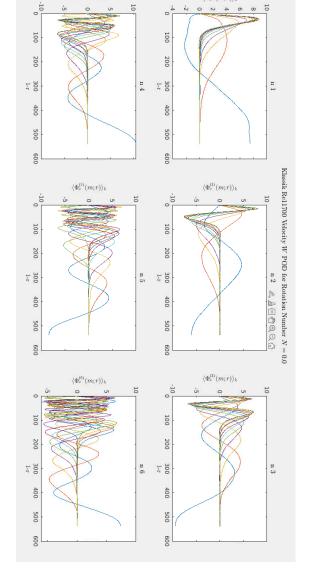
$$\lim_{\tau \to \infty} \frac{1}{\tau} \int_0^\tau \mathbf{u}_{\Gamma}(k;m;r,t) \alpha^{(n)^*}(k;m;t) \mathrm{d}t = \Phi_{\Gamma}^{(n)}(k;m;r) \lambda^{(n)}(k;m).$$

4. Result Comparison Classic/Snapshot

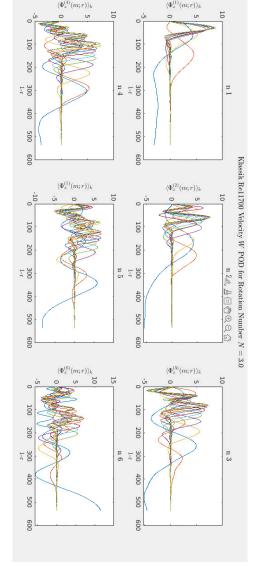


# 4.2. Snapshot-Classic Comparison

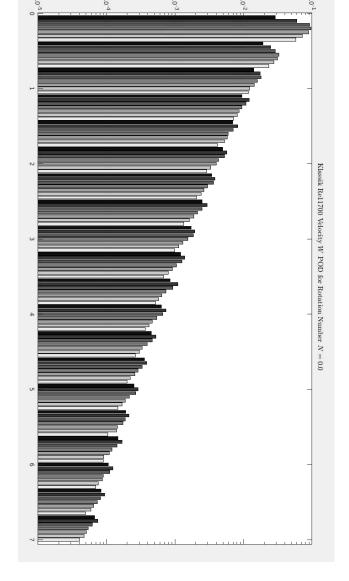


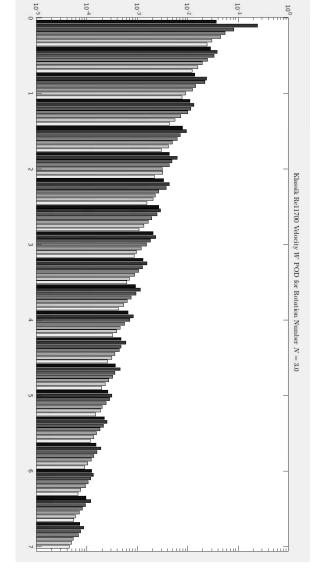


4.4. Klassik POD S=3.0



5. Energy n=0 Classic





5.2. Analysis

