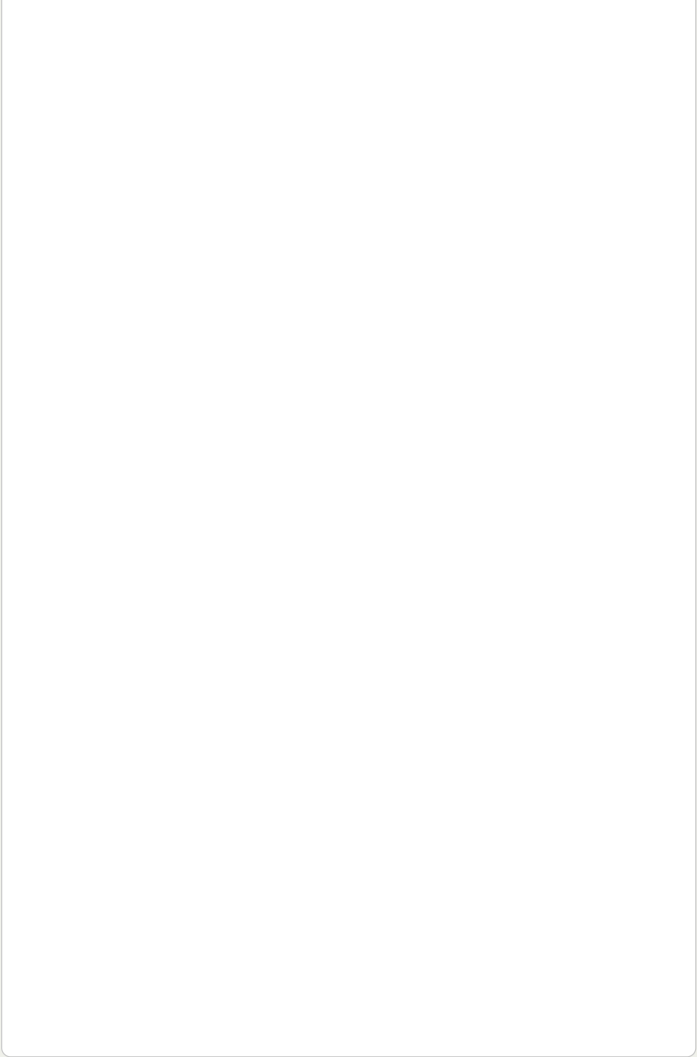


Multichamber Muffler System

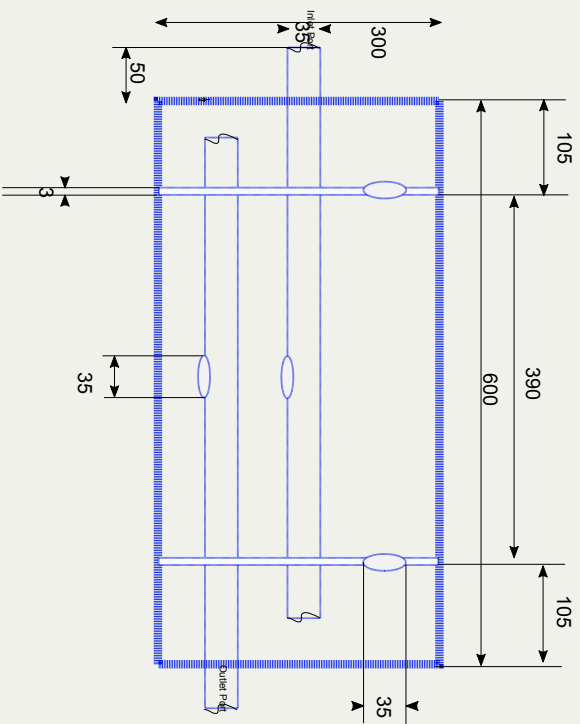
Michael Raba, MSc Candidate at University of Kentucky

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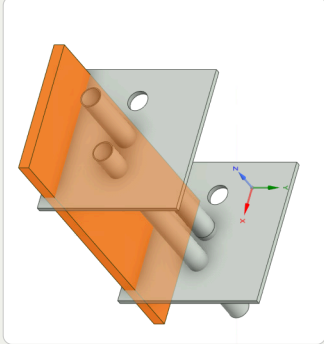


Dimensions

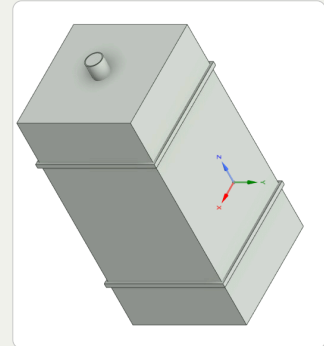
dimensional units in mm



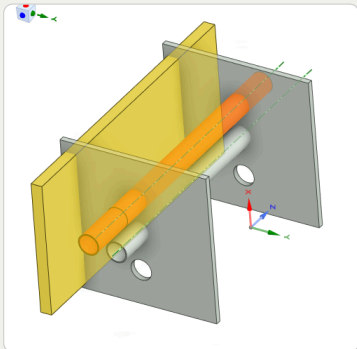
Schematic Variants for Muffler Subcomponents



Part 1 — Chamber and Baffle

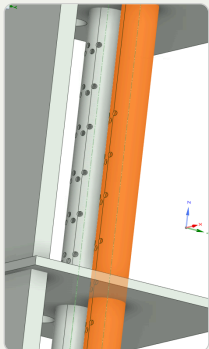


Part 2 — Fluid domain

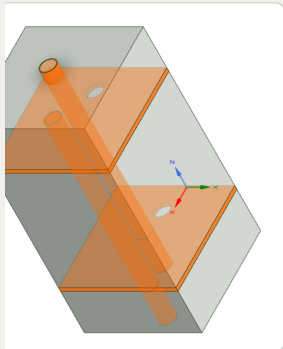


Part 3 — Fiberglass Absorbant (gold)

Part 4 — Showing perforates (aimed at fiberglass)



Part 5 — Final Assembly View



Simulated Transmission Loss (0–1000 Hz) by approximating muffler walls as fluid at 20 deg C

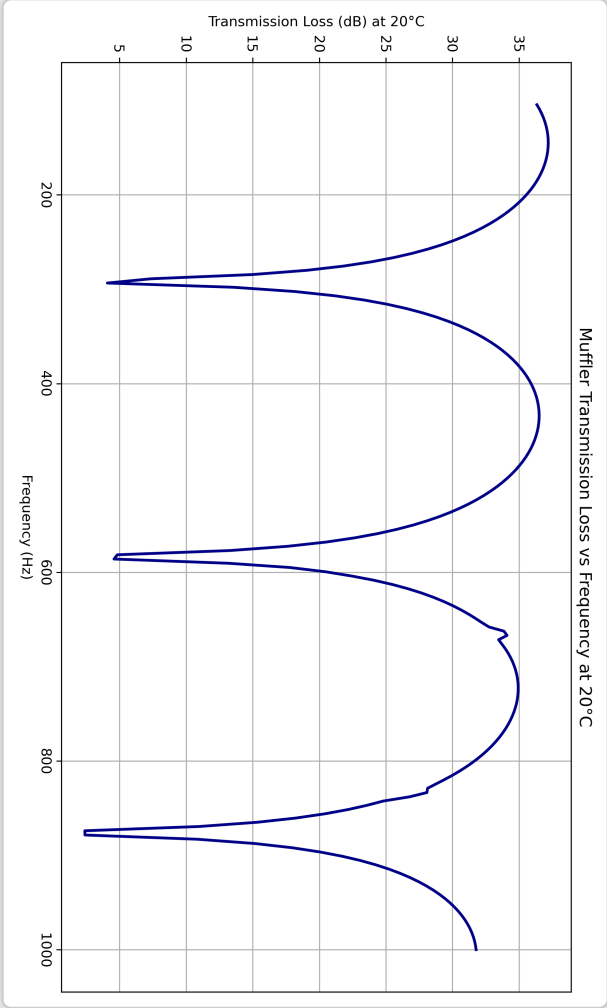
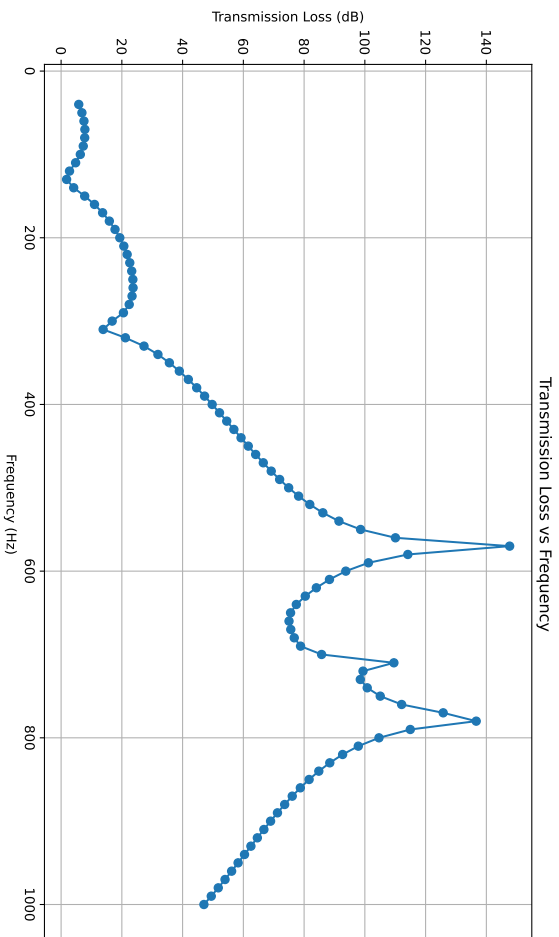


Figure: Transmission Loss curve of the muffler between 5 Hz and 1000 Hz at 20°C.

Simulab Simulation

Simulated Transmission Loss (0–1000 Hz) Simulab model



Sidlab and Ansys File Download Center

SIDLAB Model

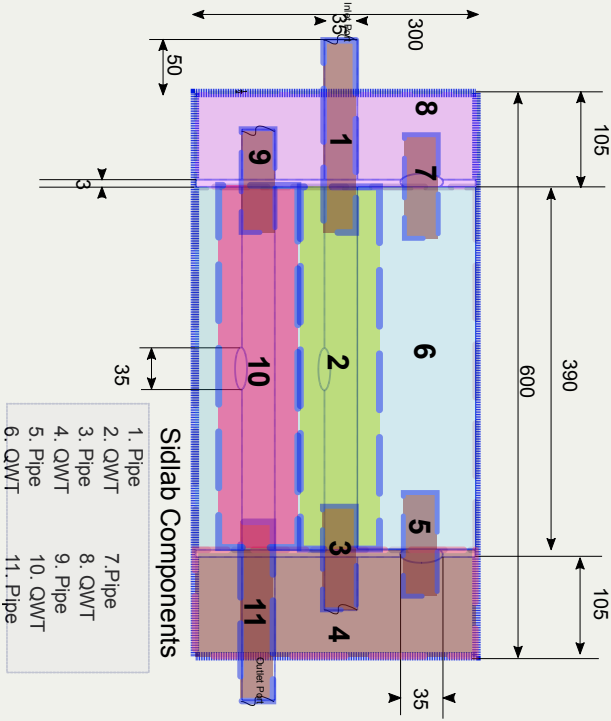
- **File:** Mark3Sid.zip
- **Created with:** SIDLAB 5.1
- [↓ Download SIDLAB File](#)

ANSYS Simulation

- **File:** Mark-I-!MDF-cleaned-data.wbpz
- **Created with:** ANSYS 2023 R2
- [↓ Download ANSYS File](#)

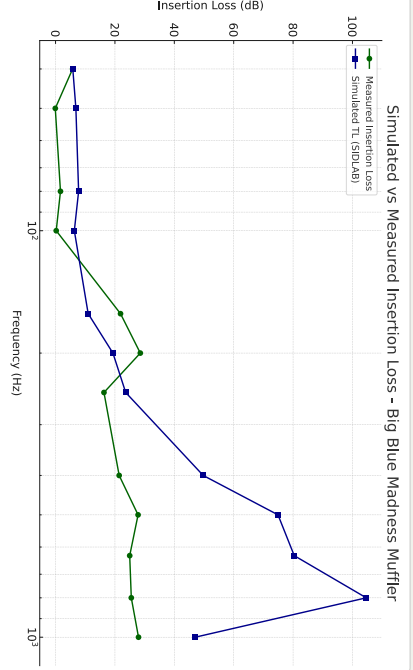
Sidlab Components

dimensional units in mm



Simulated vs Measured Insertion Loss

Measured vs Simulated TL



Insertion Loss Explanation

Insertion Loss (IL) quantifies how much sound is attenuated when a muffler is added to the system.

General formula:

$$IL = 10 \log_{10} \left(\frac{P_{baseline}}{P_{muffler}} \right)$$

Because our data is already in decibels (dB), this simplifies to:

$$IL = \text{Power}_{baseline} \text{ (dB)} - \text{Power}_{muffler} \text{ (dB)}$$

References

Cited Works

1. Munjal ML. *Acoustics of Ducts and Mufflers*. 2nd ed. Wiley; 2014. ISBN: 9781118443125.
<https://doi.org/10.1002/9781118443125>
2. Dokumaci E. *Duct Acoustics: Fundamentals and Applications to Mufflers and Silencers*. Cambridge University Press; 2021. ISBN: 9781108840750. <https://doi.org/10.1017/9781108840750>

Note: These references are foundational texts in muffler and duct acoustics and were consulted for system modeling, schematic development, and transmission loss analysis.

Anand Model: Viscoelastoplasticity and its Application to Solder Joints

Michael Raba, MSc Candidate at University of Kentucky

Created: 2025-09-15 Mon 12:39

© 1996-1997 S. J. Deane + . . .

LALIT ANAND
Massachusetts Institute of Technology
(Communicated by Theodor Lethmann, Ruhr Universität Bochum)

(Communicated by Theodor Lehmann, Ruhr Universität Bochum)

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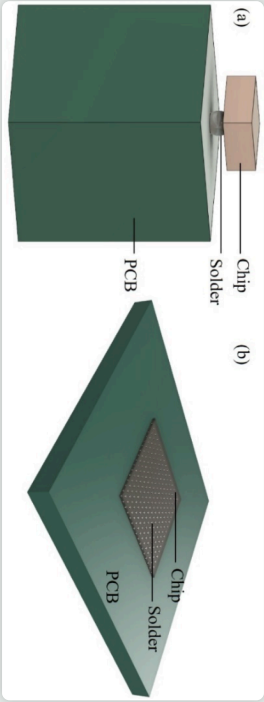
Hot-working is an important processing step during the manufacture of approximately more than eighty-five percent of all metal products. The main features of hot-working are that metals are deformed into the desired shapes at temperatures in the range of -0.5 through $-0.9 T_m$, where T_m is the melting temperature in degrees Kelvin, and at strain rates in the range of -10^{-3} through -10^3 /sec. It is to be noted that most hot-working processes are more than shape-making operations, an important goal of hot-working is to subject the workpiece to appropriate thermo-mechanical processing histories which will produce microstructures that optimize the mechanical properties of the product.

The magnitudes of metals and alloys are the physical-mechanisms of their non-isothermal conditions. The principles of the local-mechanisms of such deformation processing are now well recognized, e.g., Stevens et al. [1960], Struass & McCoy [1972], McCoy & Struass [1973], Struass [1976]. Under a deformation pass, the local-mechanisms of the material, the strain-hardening produced by the deformation tends to be counteracted by dynamic recovery processes. These recovery processes result in a rearrangement and annihilation of dislocations in such a manner that at the strain in a pass increases, the dislocation tend to arrange themselves into sub-grains walls. In some metals and alloys (especially those with a high stacking fault energy), the annihilation of dislocations tends to occur even at relatively low dislocation densities. In other metals and alloys (especially those with a low stacking fault energy), the annihilation of dislocations tends to occur only at high dislocation densities. In the former case, the recovery process is more rapid than the strain hardening process, and the material tends to be softer after the deformation pass. In the latter case, the recovery process is slower than the strain hardening process, and the material tends to be harder after the deformation pass. In other metals and alloys in which the recovery process is about the same as the strain hardening process, the material tends to be about the same after the deformation pass. In some metals with low stacking fault energies, e.g., Ni, γ-Fe and other austenitic

Case Study: Wang (2001) Apply to Solder

Why Wang's Paper Matters

- Applies Anand's unified viscoplastic framework to model solder behavior.
- Anand's model can be reduced and fitted from experiments.
- transition the theory into engineering-scale implementation.
- Targets solder joints in microelectronic packages (chip on PCB, soldered connections).

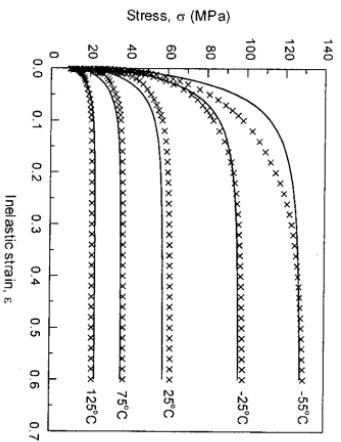


Source: Wang, C. H. (2001). "A Unified
p-Plasticity Model for Solder Alloys."
DOI: 10.1115/1.1371781

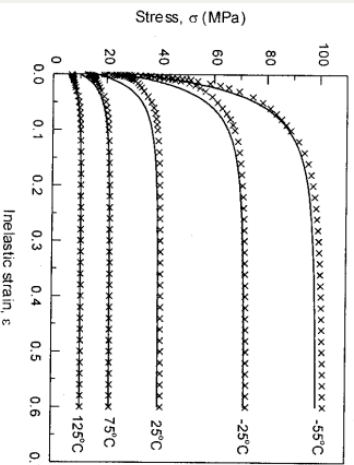
Observed Behavior

- **Top Graph (a):** $\dot{\epsilon} = 10^{-2} \text{ s}^{-1}$
- High strain rate \rightarrow higher stress
- Recovery negligible \rightarrow pronounced hardening
- **Bottom Graph (b):** $\dot{\epsilon} = 10^{-4} \text{ s}^{-1}$
- Lower strain rate \rightarrow lower stress at same strain
- Recovery and creep effects more significant

Model Accuracy: Lines = model prediction, X = experimental data



(a) $\dot{\epsilon} = 1.0 \times 10^{-2} \text{ s}^{-1}$



(b) $\dot{\epsilon} = 1.0 \times 10^{-4} \text{ s}^{-1}$

Key Insights from Wang (2001)

- "At lower strain rates, recovery dominates... the stress levels off early."
- "At high strain rates, hardening dominates, and the stress grows continuously."

Anand's model smoothly captures strain-rate and temperature dependence of solder materials.

Main Equations of Wang's Anand-Type Viscoplastic Model

Flow Rule (Plastic Strain Rate)

- $\dot{\varepsilon}^p = A \exp\left(-\frac{Q}{RT}\right) \left[\sinh\left(\frac{j\sigma}{s}\right)\right]^{1/m}$
- Plastic strain rate increases with stress and temperature.
- No explicit yield surface; flow occurs at all nonzero stresses.

Deformation Resistance Saturation s^*

- $s^* = \hat{s} \left(\frac{\dot{\varepsilon}^p}{A} \exp\left(\frac{Q}{RT}\right) \right)^n$
- Defines the steady-state value that s evolves toward.
- Depends on strain rate and temperature.

Evolution of Deformation Resistance s

- $\dot{s} = h_0 \left| 1 - \frac{s}{s^*} \right|^a \operatorname{sign}\left(1 - \frac{s}{s^*}\right) \dot{\varepsilon}^p$
- Describes dynamic hardening and softening of the material.
- s evolves depending on proximity to s^* and flow activity.

Note: Constants $A, Q, m, j, h_0, \hat{s}, n, a$ are material-specific and fitted to experimental creep/strain rate data.

Image Reference

Values are from correspond to 60Sn40Pb solder parameters used in Anand's model:

- S_0 : Initial deformation resistance
- Q/R : Activation energy over gas constant
- A : Pre-exponential factor for flow rate
- ξ : Multiplier of stress inside sinh
- m : Strain rate sensitivity of stress
- h_0 : Hardening/softening constant
- \hat{s} : Coefficient for saturation stress
- n : Strain rate sensitivity of saturation
- a : Strain rate sensitivity of hardening or softening

Numerical Values

- $S_0 = 5.633 \times 10^7 \text{ Pa}$
- $Q/R = 10830 \text{ K}$
- $A = 1.49 \times 10^7 \text{ s}^{-1}$
- $\xi = 11$
- $m = 0.303$
- $h_0 = 2.6408 \times 10^9 \text{ Pa}$
- $\hat{s} = 8.042 \times 10^7 \text{ Pa}$
- $n = 0.0231$
- $a = 1.34$

These constants match Wang's paper for modeling 60Sn40Pb viscoplasticity.

Forward Euler Explicit time integration scheme Pseudocode

Initialization

- Material constants: $A, Q/R, j, m, h_0, \hat{s}, n, a, E$
- Strain rate: $\dot{\epsilon}$
- Temperature set: $\{T_i\}$
- Set: $\epsilon^P(0) = 0, \quad s(0) = \hat{s}$

Time Evolution Loop

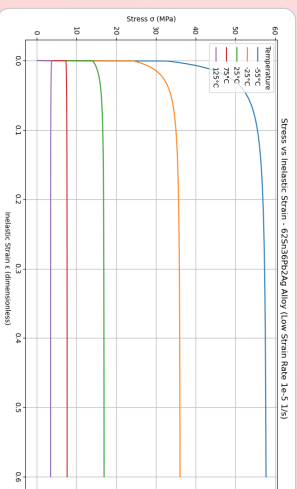
1. $\epsilon_{\text{total}}(t) = \dot{\epsilon} t$
2. $\sigma_{\text{trial}} = E(\epsilon_{\text{total}} - \epsilon^P)$
3. Compute $x = \frac{j\sigma}{s}$
4. Approximate $\sinh(x)$ (linearize if $|x| \ll 1$)
5. $\epsilon^P = Ae^{-Q/RT}(\sinh(x))^{1/m}$

Plastic Flow & Resistance Evolution

6. $s^* = \hat{s} \left(\frac{\dot{\epsilon}^P}{A} e^{Q/RT} \right)^n$
7. $\dot{s} = h_0 \left| 1 - \frac{s}{s^*} \right|^a \text{sign} \left(1 - \frac{s}{s^*} \right) \dot{\epsilon}^P$
8. Update: $\epsilon^P(t + \Delta t) = \epsilon^P(t) + \dot{\epsilon}^P \Delta t$
9. Update: $s(t + \Delta t) = s(t) + \dot{s} \Delta t$
10. Record $(\epsilon_{\text{total}}, \sigma_{\text{trial}})$

Termination

- Stop when $\epsilon_{\text{total}} \geq \epsilon_{\text{max}}$
- Plot σ vs ϵ for all T_i



Forward Euler Scheme for Anand Model

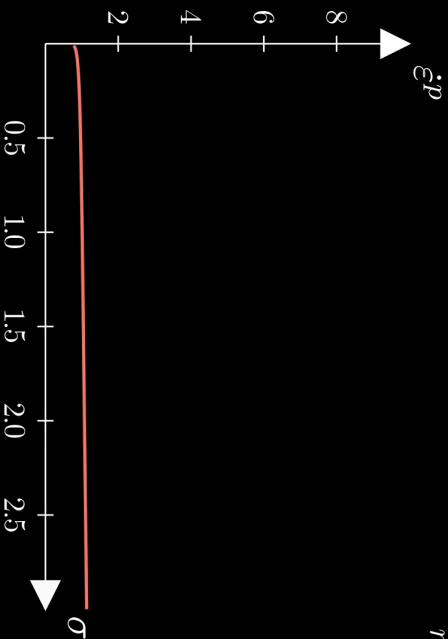
[illegible]

Strain rate sensitivity of stress m

- As $m \rightarrow 0$, rate insensitive (yield)
- As $m \rightarrow 1$, small stress change causes big change in strain rate

Anand Flow Law: Varying m

$$m = 0.05$$



Tensorial Flow Rule (directional form)

$$\mathbf{D}^p = \dot{\epsilon}^p \left(\frac{3}{2} \frac{\mathbf{T}'}{\bar{\sigma}} \right)$$

Equivalent Stress Definition

$$\bar{\sigma} = \sqrt{\frac{3}{2} \mathbf{T}' : \mathbf{T}'}$$

Plastic Strain Rate (magnitude form)

$$\dot{\epsilon}^p = A \exp \left(-\frac{Q}{R\theta} \right) \left[\sinh \left(\xi \frac{\bar{\sigma}}{s} \right) \right]^{1/m}$$

Full Flow Rule with Hyperbolic Sine

$$\begin{aligned} \mathbf{D}^p &= A \exp \left(-\frac{Q}{R\theta} \right) \left[\sinh \left(\xi \frac{\bar{\sigma}}{s} \right) \right]^{1/m} \left(\frac{3}{2} \frac{\mathbf{T}'}{\bar{\sigma}} \right), \\ &= \dot{\gamma}^p \left(\frac{\tilde{\mathbf{T}}'}{2\tilde{\tau}} \right), \quad \tilde{\tau} = \left\{ \frac{1}{2} \text{tr}(\tilde{\mathbf{T}}'^2) \right\}^{1/2} \end{aligned}$$

- Direction given by \mathbf{T}' .
- Magnitude determined by hyperbolic sine based on $\bar{\sigma}/s$.
- $\tilde{\tau}$ represents the effective shear stress computed from deviatoric stress.
- $\bar{\sigma} = \sqrt{\frac{3}{2} \mathbf{T}' : \mathbf{T}'}$ is the von Mises Equivalent stress, but is formally defined without yield point
- Full flow = **direction** × **magnitude**.

Summary:

Evolution Equation for the Stress

Stress Evolution Equation (Rate form of Hooke's Law)

Material Tensors and Operators

$$\overset{\nabla}{\mathbf{T}} = \mathbb{L}[\mathbf{D} - \mathbf{D}^p] - \mathbf{\Pi}\dot{\theta}$$

Rate-form Hooke's law for finite deformation plasticity, with frame-indifference enforced through the Jaumann rate.)

Jaumann Rate Definition

$$\overset{\nabla}{\mathbf{T}} = \dot{\mathbf{T}} - \mathbf{W}\mathbf{T} + \mathbf{T}\mathbf{W}$$

- $\mathbb{L} = 2\mu\mathbf{I} + (\kappa - \frac{2}{3}\mu)\mathbf{1} \otimes \mathbf{1}$ — isotropic elasticity tensor
- $\mathbb{L}\mathbf{D}$ represents how instantaneous strain rates generate stresses according to the elastic material's stiffness properties.
- $\mu = \mu(\theta)$, $\kappa = \kappa(\theta)$ — temperature-dependent moduli
- $\mathbf{\Pi} = (3\alpha\kappa)\mathbf{1}$ — stress-temperature coupling
- $\alpha = \alpha(\theta)$ — thermal expansion coefficient
- $\mathbf{D} = \text{sym}(\nabla \mathbf{v})$ — stretching tensor
- $\mathbf{W} = \text{skew}(\nabla \mathbf{v})$ — spin tensor
- \mathbf{I} = fourth-order identity tensor
- $\mathbf{1}$ = second-order identity tensor

- Summary:**
- Stress rate follows Jaumann derivative to ensure frame indifference.
 - Elastic response governed by isotropic fourth-order tensor \mathbb{L} .
 - Thermal expansion introduces additional stress through $\mathbf{\Pi}\dot{\theta}$.

Stress Evolution and Thermal Effects

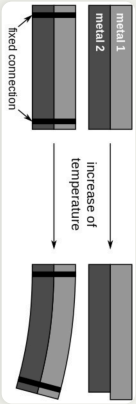
In the stress evolution equation,

$$\dot{\mathbf{T}} = \mathbb{L}[\mathbf{D} - \mathbf{D}^p] - \mathbf{\Pi}\dot{\theta},$$

term $\mathbf{\Pi}\dot{\theta}$ represents the stress change that would occur due to pure thermal expansion alone, without any mechanical loading.

Why Subtract the Thermal Term?

- Thermal expansion creates strain even without external forces.
- Without subtracting $\mathbf{\Pi}\dot{\theta}$, the model would falsely attribute thermal strain as mechanical stress.
- Subtracting isolates the true mechanical response from thermal effects.



Summary:

- Thermal expansion induces strain without force.
- Subtracting $\mathbf{\Pi}\dot{\theta}$ ensures only mechanical strains generate stresses.
- This keeps the constitutive model physically accurate during heating and cooling.

Relaxed (Intermediate) Configuration

Context for the Relaxed Configuration

The relaxed configuration represents the material after removing plastic deformations but before applying new elastic deformations.

It is introduced to separate permanent plastic effects from recoverable elastic effects.

All thermodynamic potentials, internal variables, and evolution laws are defined relative to this frame.

The relaxed state provides a clean, natural reference for measuring elastic strain E^e and computing dissipation.

What Happens in the Relaxed Configuration?

- The elastic deformation gradient F^e is measured from the relaxed state to the current deformed state.

- Elastic strain measures like C^e and E^e are defined in this configuration.

- The Kirchhoff stress $\bar{\mathbf{T}}$ is naturally associated with the relaxed volume.

- Plastic flow is accounted for separately through the plastic velocity gradient \mathbf{L}^p .

Summary:

- The relaxed configuration isolates elastic responses cleanly, enabling proper definition of thermodynamics and plastic evolution laws.

Kinemematics in the Relaxed Configuration

Stress and Power Quantities

- Elastic deformation gradient:

$$F = F^e F^p \Rightarrow F^e = F F^{p-1}$$

- Kirchhoff stress (weighted Cauchy stress):

$$\tilde{\mathbf{T}} = (\det F) \mathbf{T}$$

- Elastic right Cauchy-Green tensor:

$$C^e = F^{eT} F^e$$

- Stress power split:

$$\dot{w} = \dot{w}^e + \dot{w}^p$$

- Elastic Green–Lagrange strain tensor:

$$E^e = \frac{1}{2}(C^e - I)$$

$$\dot{w}^e = \tilde{\mathbf{T}} : \dot{E}^e, \quad \dot{w}^p = (C^e \tilde{\mathbf{T}}) : \mathbf{L}^p$$

Summary:

- Elastic kinematics and stress measures are formulated relative to the relaxed configuration, cleanly separating plastic and elastic contributions.
- Stress Power Split allows us to cleanly isolate plastic dissipation from elastic storage.
- Green–Lagrange strain tensor E^e is used because it symmetrically captures nonlinear elastic strain relative to the relaxed configuration
- The right Cauchy–Green tensor $C^e = F^{eT} F^e$ is required as an intermediate to compute E^e from the elastic deformation gradient F^e without referencing spatial coordinates

Thermodynamic Separation

1. Start with Total Dissipation:

$$\mathcal{D} = \dot{\omega} - \dot{\psi} \geq 0$$

where $\dot{\omega} = \hat{\mathbf{T}} : \dot{\mathbf{E}}^e + (C^e \hat{\mathbf{T}}) : \mathbf{L}^p$

2. Split Stress Power:

$$\dot{\omega} = \dot{\omega}^e + \dot{\omega}^p$$

with:

- $\dot{\omega}^e = \hat{\mathbf{T}} : \dot{\mathbf{E}}^e$
- $\dot{\omega}^p = (C^e \hat{\mathbf{T}}) : \mathbf{L}^p$

3. Group Terms with $\dot{\psi}$:

$$(\dot{\omega}^e - \dot{\psi}) + \dot{\omega}^p \geq 0$$

4. Apply Elastic Energy Consistency:

$$\dot{\omega}^e - \dot{\psi} = 0 \quad \Rightarrow \quad \dot{\omega}^p \geq 0$$

Key Physical Insights

- **Elastic deformations** are recoverable and do not cause entropy production.
- **All dissipation** stems from the plastic flow: $\dot{\omega}^p$.
- **Plastic work** increases entropy and governs viscoplastic evolution.

Summary:

The stress power split ensures that the second law is satisfied by assigning dissipation solely to irreversible processes.

Reference Configuration

Framework in the Reference Configuration

Key Equations in the Reference Frame

The free energy ψ is defined relative to the reference configuration.

State variables like E^e , θ , \bar{g} , $\bar{\mathbf{B}}$, s are used as arguments of ψ .

Stress is expressed using the second Piola–Kirchhoff tensor \mathbf{S} .

Dissipation inequality, stress–strain relations, and evolution laws are all written in reference variables.

Mass density ρ_0 from the reference configuration normalizes all terms.

• Free energy:

$$\psi = \psi(E^e, \theta, \bar{g}, \bar{\mathbf{B}}, s)$$

• Dissipation inequality:

$$\dot{\psi} + \eta \dot{\theta} - \rho_0^{-1} \mathbf{S} : \dot{\mathbf{E}} + (\rho_0 \theta)^{-1} \mathbf{q}_0 \cdot \dot{\mathbf{g}}_0 \cdot \mathbf{g}_0 \leq 0$$

• Constitutive relation:

$$\mathbf{S} = \rho_0 \frac{\partial \psi}{\partial E^e}$$

Summary:

- In the reference configuration, all energy storage, stress updates, and internal variable evolution are formulated with reference-frame quantities for consistency and objectivity.

Thermodynamic Quantities

- Free energy density:

$$\boxed{\psi = \epsilon - \theta \eta}$$

- Reduced dissipation inequality:

$$\boxed{\dot{\psi} + \eta \dot{\theta} - \rho^{-1} \mathbf{T} : \mathbf{L} + (\rho \theta)^{-1} \mathbf{q} \cdot \mathbf{g} \leq 0}$$

- State variables:

$$\{\bar{E}^e, \theta, \bar{g}, \bar{\mathbf{B}}, s\}$$

with E^e as elastic strain and s as internal resistance.

Stress Power and Kirchhoff Stress

- Stress power per relaxed volume:

$$\boxed{\dot{\omega} = \left(\frac{\rho_0}{\rho} \right) \mathbf{T} : \mathbf{L}}$$

- Weighted Cauchy (Kirchhoff) stress:

$$\boxed{\tilde{\mathbf{T}} = (\det F) \mathbf{T}}$$

or

$$\boxed{\tilde{\mathbf{T}} = \left(\frac{\rho_0}{\rho} \right) \mathbf{T}}$$

- Decomposition of stress power:

$$\boxed{\dot{\omega} = \dot{\omega}^e + \dot{\omega}^p}$$

$$\dot{\omega}^e = \tilde{\mathbf{T}} : \dot{\bar{E}}^e, \quad \dot{\omega}^p = (C^e \tilde{\mathbf{T}}) : \mathbf{L}^p$$

- Summary:**
- Free energy and dissipation govern thermodynamic consistency,
 - Stress power naturally splits into elastic and plastic parts,
 - Kirchhoff stress simplifies stress evolution accounting for volume changes.

POD Analysis of Turbulent Pipe Flow

M. Raba

Created: 2025-09-15 Mon 12:39

1. Code Execution and Layout

1.1. Layout

1. b7.m
2. initSpectral.m
 - reads in binary files, takes eg m-fft
3. \hookrightarrow initEigs.m
 - forms corrMat, finds eigenvalues

1. \hookrightarrow initPod.m
- 1.2. Layout 2
- carries out POD calculations (quadrature, multiplication \mathbf{e}_θ^T between $\alpha\Phi$) according to Papers (Citriniti George 2000 for Classic POD, Hellsrom Smit 2017 for Snapshot POD)
2. \hookrightarrow timeReconstructFlow.m
- performs 2d reconstruction + plotSkmr (generates 1d radial graph)

1.3. Important Switches

`pipe = Pipe()`; creates a Pipe Class. As the functions (above) are called, data is stored in sub-structs:

1. `obj.CaseId` - stores properties like Re, rotation number S , experimental flags such as quadrature (simpson/trapezoidal), number of gridpoints, frequently called vectors ($rMat\ r = 1, \dots, 0.5$)
2. `obj.pod` - eigen data, used for calculating POD
3. `obj.solution` - computed POD modes
4. `obj.plt` - plot configuration

2. Equations Used in Code Procedure

2.1. Classic POD Equations

The following equations are used in the above code.

$$\begin{aligned}\int_{r'} \mathbf{S}(k; m; r; r') \Phi^{(n)}(k; m; r') r' dr' &= \lambda^{(n)}(k; m) \Phi^{(n)}(k; m; r) \\ \mathbf{S}(k; m; r; r') &= \lim_{\tau \rightarrow \infty} \frac{1}{\tau} \int_0^\tau \mathbf{u}(k; m; r; t) \mathbf{u}^*(k; m; r', t) dt \\ \alpha^{(n)}(k; m; t) &= \int_r \mathbf{u}(k; m; r; t) \Phi^{(n)*}(k; m; r) r dr\end{aligned}$$

2.2. Classic POD Equations (Fixed)

$$\begin{aligned}
& \int_r r^{1/2} S_{i,j}(r, r^i; m; f) r^{1/2} \phi_j^{*(n)}(r^i; m; f) r^{1/2} dr \\
&= \underbrace{\lambda^{(n)}(m, f)}_{\lambda^{(n)}(m; f)} \underbrace{r^{1/2} \phi_i^{(n)}(r; m; f)}_{\phi_i^{(n)}(r; m; f)} \underbrace{\phi_j^{*(n)}(r^i; m; f)}_{\phi_j^{*(n)}(r^i; m; f)} \\
&\alpha_n(m; t) = \int_r \mathbf{u}(m; r, t) r^{1/2} \Phi_n^*(m; r) dr
\end{aligned}$$

2.3. Snapshot POD Equations

$$\begin{aligned}
& \lim_{\tau \rightarrow \infty} \frac{1}{\tau} \int_0^\tau \mathbf{u}_\mathbf{r}(k; m; r, t) \alpha^{(n)*}(k; m; t) dt \\
&= \Phi_\mathbf{r}^{(n)}(k; m; r) \lambda^{(n)}(k; m) \\
\mathbf{R}(k; m; t, t') &= \int_r \mathbf{u}(k; m; r, t) \mathbf{u}^*(k; m; r, t') r dr \\
& \lim_{\tau \rightarrow \infty} \frac{1}{\tau} \int_0^\tau \mathbf{u}_\mathbf{r}(k; m; r, t) \alpha^{(n)*}(k; m; t) dt \\
&= \Phi_\mathbf{r}^{(n)}(k; m; r) \lambda^{(n)}(k; m).
\end{aligned}$$

2.4. Reconstruction

The reconstruction is given by

$$q(\xi, t) - \bar{q}(\xi) \approx \sum_{j=1}^r a_j(t) \varphi_j(\xi) \Rightarrow$$

$$q(r, \theta, t; x) = \bar{q}(r, \theta, t; x) + \sum_{n=1} \sum_{m=0} \alpha^{(n)}(m; t) \Phi^{(n)}(r; m; x)$$

Since the snapshot pod implementation is not error-free, the reconstruction can only be recovered by writing for $\text{factor} \gg 0$.

$$q(r, \theta, t; x) = \bar{q}(r, \theta, t; x) + (\text{factor } \gamma) \sum_{n=1} \sum_{m=0} \alpha^{(n)}(m; t) \Phi^{(n)}(r; m; x)$$

2.5. Reconstruction

In order to reconstruct in code, `caseId.fluctuation = 'off'`. This is incorrect. The necessary use of (factor γ) is incorrect

3. Derivation

To derive the questioned equation, consider the integral:

$$\frac{1}{\tau} \int_0^\tau \mathbf{u}_\Gamma(k; m; \tau, t) \alpha^{(n)*}(k; m; t) dt.$$

Substitute \mathbf{u}_Γ with its expansion:

$$\frac{1}{\tau} \int_0^\tau \left(\sum_l \Phi_{\mathbf{I}}^{(l)}(k; m; \tau) \alpha^{(l)}(k; m; t) \right) \alpha^{(n)*}(k; m; t) dt.$$

3.1.4 Derivation

Exchange the order of summation and integration, and apply orthogonality,

$$\sum_l \Phi_{\mathbf{T}}^{(l)}(k; m; r) \left(\frac{1}{\tau} \int_0^\tau \alpha^{(l)}(k; m; t) \alpha^{(n)*}(k; m; t) dt \right).$$

Due to the orthogonality, namely that $\alpha^{(n)}$ and $\alpha^{(p)}$ are uncorrelated

$$\langle \alpha^{(n)} \alpha^{(p)} \rangle = \lambda^{(n)} \delta_{np}$$

all terms where $l \neq n$ will vanish, and there remains only the $l = n$ term,

$$\Phi_{\mathbf{T}}^{(n)}(k; m; r) \left(\frac{1}{\tau} \int_0^\tau \alpha^{(n)}(k; m; t) \alpha^{(n)*}(k; m; t) dt \right).$$

derivation assumes the normalization of modes and their orthogonality, along with the eigenvalue relationship to simplify the original integral into a form that reveals the spatial structure ($\Phi_{\mathbf{T}}^{(n)}$) of each mode scaled by its significance ($\lambda^{(n)}$).

3.2. 6 Derivation

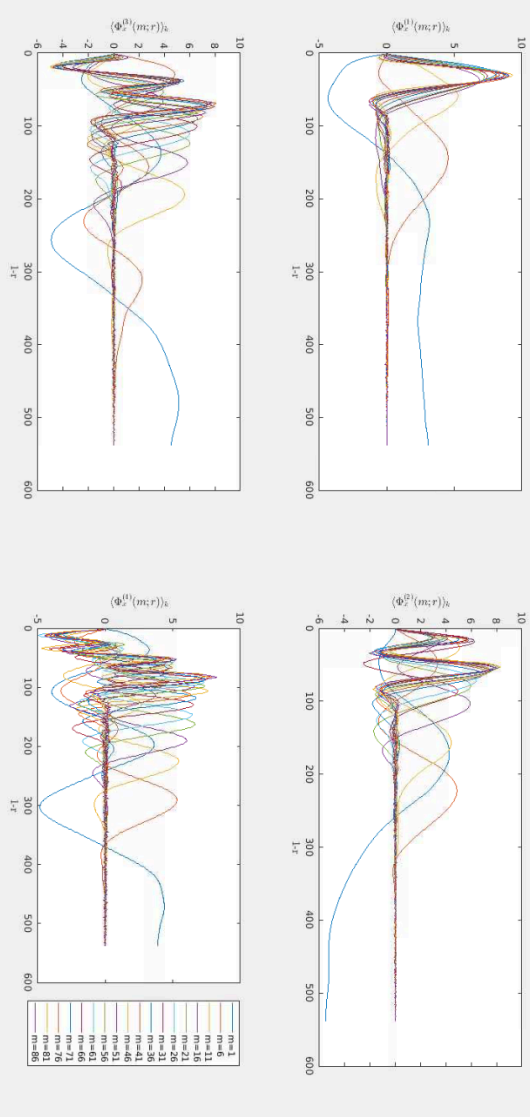
Cross-correlation tensor \mathbf{R} is defined as $\mathbf{R}(k; m; t, t') = \int_r \mathbf{u}(k; m; r, t) \mathbf{u}^*(k; m; r, t') r \, dr$. This tensor is now transformed from $[3r \times 3r']$ to a $[t \times t']$ tensor. The n POD modes are then constructed as,

$$\lim_{T \rightarrow \infty} \frac{1}{T} \int_0^T \mathbf{u}_n(k; m; r, t) \alpha^{(n)*}(k; m; t) dt = \Phi_n^{(n)}(k; m; r) \lambda^{(n)}(k; m).$$

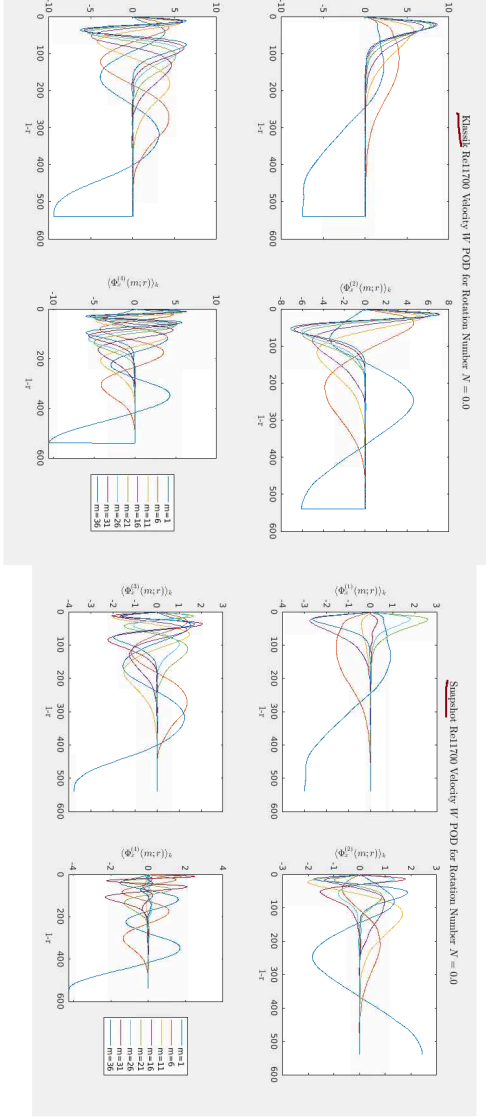
4. Result Comparison Classic/Snapshot

4.1. Radial Classic

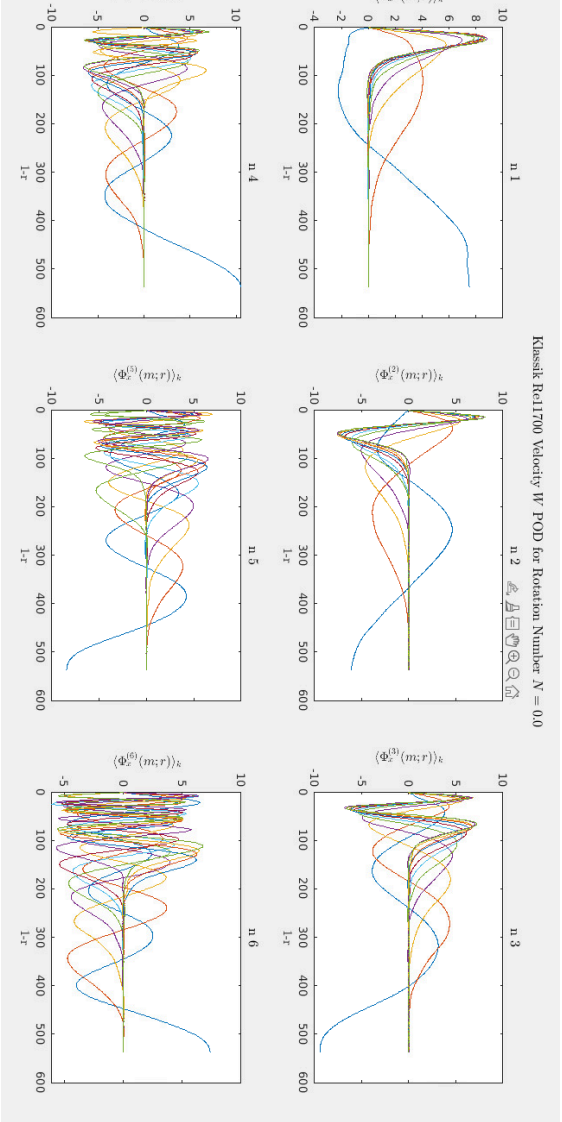
Klasik Re1170 Velocity w POD for Rotation Number $N = 3.0$



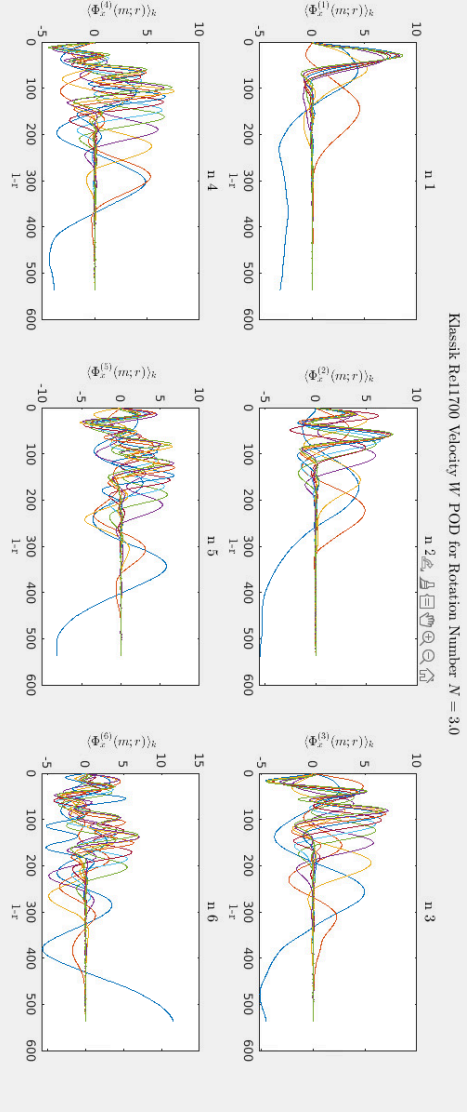
4.2. Snapshot-Classic Comparison



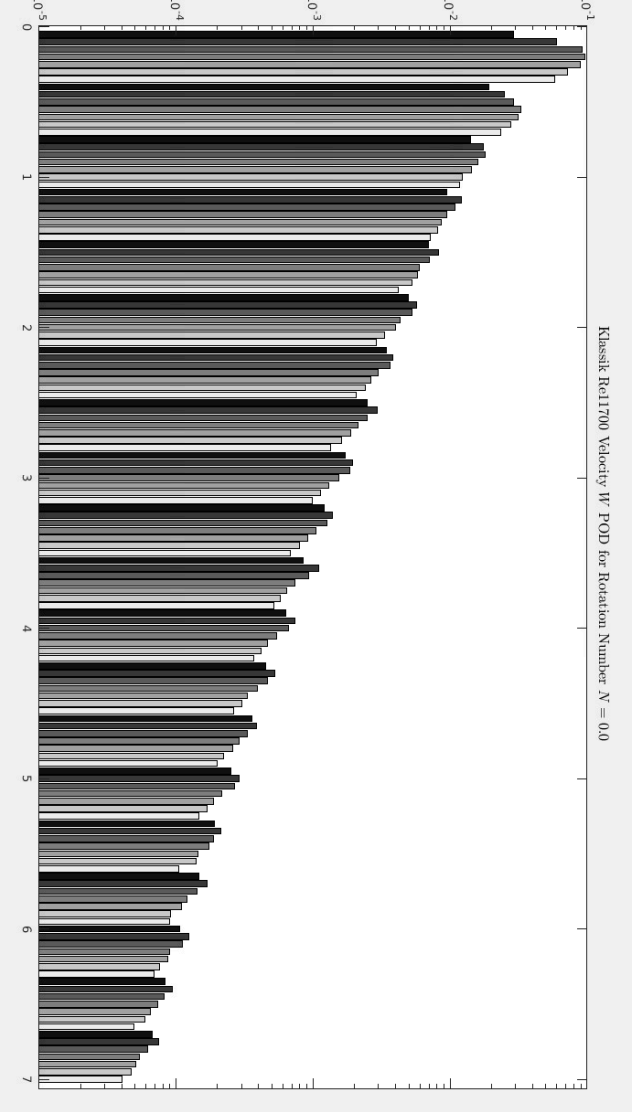
4.3. Klassik POD S=0.0



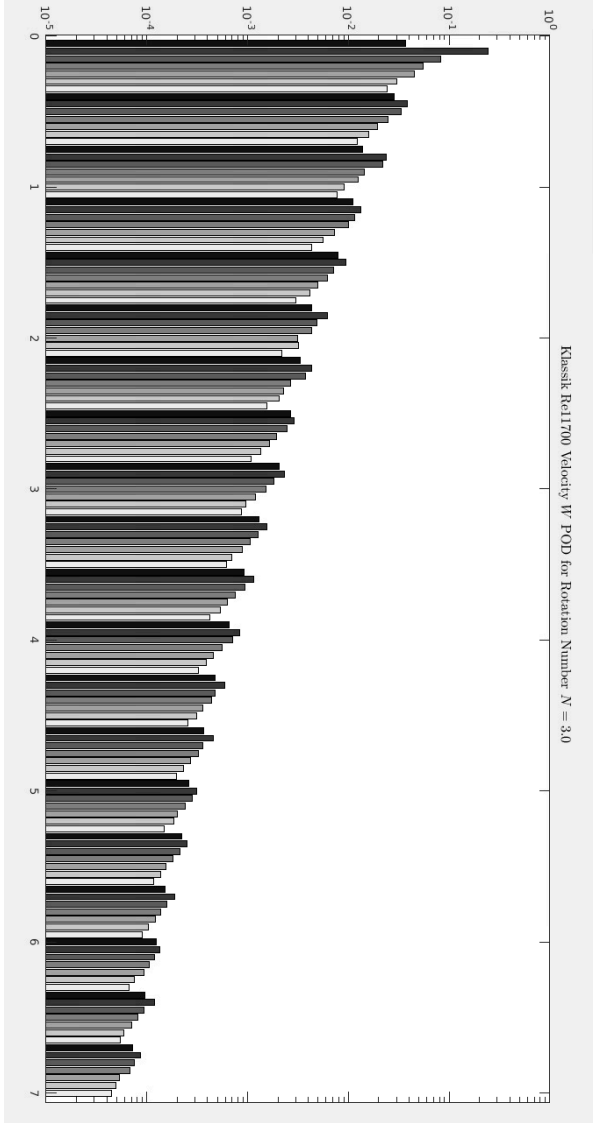
4.4. KLASIK POD S=3,0



5. Energy n=0 Classic



5.1. $n=3$ Classic



5.2. Analysis

6. Reconstruction

6.1. Reconstruction

